

# Bracket — Fatigue Evaluation

# Introduction

The S-N curve, also called the Wöhler curve, is one of the most popular methods for fatigue evaluation. The curve relates the stress amplitude to the limiting fatigue life and can be obtained directly from a set of standard fatigue test. Often, a structure is however subjected to conditions different from the experimental conditions of the fatigue tests. The fatigue data must then be appropriately modified in order to take the actual operating conditions into account.

In this example it is shown how to perform a fatigue evaluation when the material data needs to account for harsh environmental conditions and poor manufacturing process.

# Model Definition

The bracket geometry can be seen in Figure 1. The component is subjected to external loads that lift one arm up and pull the other arm down. The load magnitude is cycled between a zero load, a peak load and back to zero load. Additional information regarding the model set up can be found in the documentation of the application *Bracket-Spring Foundation Analysis*, found in the Structural Mechanics Module.



Figure 1: Bracket geometry.

The structural steel in the bracket has poor cleanliness and contains fairly large inclusions. It is well known that with decreasing purity the endurance limit decreases and thus large inclusions shorten the lifetime. The fatigue properties of the material have been tested in a material laboratory and are summarized in Table 1. The material has a distinct endurance

limit at 110 MPa. Tests were performed in nominal testing conditions and on specimens with a very good surface finish.

Fatigue lifetime (cycles)	Stress amplitude (MPa)
1.10e3	360
2.17e3	320
3.82e3	290
7.15e3	260
l.45e4	230
3.23e4	200
5.92e4	180
l.16e5	160
2.51e5	140
6.09e5	120
1.00e6	110

TABLE I: S-N CURVE DATA.

As opposed to the testing conditions, the bracket was machined and has a rough surface as a result. Moreover the bracket operates in salt water that has a strong influence on the fatigue resistance. Since there is lack of test data for the operating conditions, the S-N curve data must be modified. One approach is to use

$$\sigma_a = k \cdot f_{\rm SN}(N) \tag{1}$$

where  $\sigma_a$  is the stress amplitude, *k* is the modification factor, *N* is the fatigue lifetime, and  $f_{\rm SN}$  is the S-N curve. Based on experience, the modification factor for the corrosion in salt water can be set to 0.4. The modification factor for the surface finish, due to machining manufacturing technique, can be set to 0.7. The modification factor in Equation 1 for the combined effect is simply the product of the modification factors for all operating conditions and thus 0.28.

# Results and Discussion

The stress distribution in the bracket at the peak load is shown in Figure 2.



Figure 2: Stress in the bracket at the peak operating load.

A close-up of the part with stress concentrations reveals that stresses are significant also on the inner side of the bracket, see Figure 3.



Figure 3: A close-up of the part of the bracket that experiences the highest stresses.

The fatigue analysis of the bracket material, modified for the manufacturing conditions predicts infinite life. When submerged in saltwater, the stresses in most of the bracket are not high enough to cause fatigue. Only in a few local points is there is a risk of fatigue, see



Figure 4. The bracket most probably fails in the connection between the two flat parts since the stresses on both the upper and the lower sides are high enough to initiate fatigue.

Figure 4: Fatigue lifetime when bracket is submerged in salt water.

In Figure 4 a finite life is predicted close to the holes that fasten the bracket. Those results are highly dependent on the boundary conditions that were applied there. In this model a spring foundations was used to fasten the bracket, but the constraint could have also been applied using a fixed connection of with a full contact analysis. This would change the stress field in the vicinity of the holes and result in a different fatigue lifetime.

# Notes About the COMSOL Implementation

The S-N curve can be defined using different function types in COMSOL Multiphysics. When using an interpolation function it is important to use many points to specify the relation between fatigue life and stress amplitude, since the range of the fatigue life is very large and a small change in the stress amplitude results in a large change in the fatigue life. The function value between two data points can be evaluated in different ways but linear interpolation is the most common one. In Figure 5 the S-N curve is specified with eleven and with four measurement points respectively. Note that the fatigue life is displayed in a logarithmic scale and therefore the interpolation is not a straight line. The two curves

display large differences. At 200 MPa one curve predicts  $3.23 \cdot 10^4$  cycles while the other one predicts  $5.80 \cdot 10^4$  cycles; a difference by almost a factor two.



Figure 5: S-N curve based different number of measurement points.

**Application Library path:** Fatigue\_Module/Stress\_Life/bracket\_fatigue

# Modeling Instructions

## ROOT

In this example you will start from an existing model from the Structural Mechanics Module.

## APPLICATION LIBRARIES

- I From the File menu, choose Application Libraries.
- 2 In the Application Libraries window, select Structural Mechanics Module>Tutorials> bracket\_spring in the tree.
- 3 Click 💿 Open.

#### ROOT

Fatigue is calculated based on a load cycle. Create a parameter to scale the magnitude of the external load.

## **GLOBAL DEFINITIONS**

#### Parameters 1

- I In the Model Builder window, under Global Definitions click Parameters I.
- 2 In the Settings window for Parameters, locate the Parameters section.
- **3** In the table, enter the following settings:

Name	Expression	Value	Description
para	0	0	Load cycle control
k	1.0	1	Stress factor

#### COMPONENT I (COMPI)

In the Model Builder window, expand the Component I (compl) node.

## DEFINITIONS

#### Analytic I (load)

- I In the Model Builder window, expand the Component I (compl)>Definitions node, then click Analytic I (load).
- 2 In the Settings window for Analytic, locate the Definition section.
- 3 In the Expression text field, type para\*F\*cos(atan2(py,abs(px))).

Start by improving the existing mesh in order to get a better resolution of stress gradients at the critical fillets.

# MESH I

In the Model Builder window, expand the Component I (compl)>Mesh I node.

## Size 2

- I In the Model Builder window, expand the Component I (compl)>Mesh I> Free Tetrahedral I node.
- 2 Right-click Free Tetrahedral I and choose Size.
- 3 In the Settings window for Size, locate the Geometric Entity Selection section.
- **4** From the **Geometric entity level** list, choose **Boundary**.
- 5 Select Boundaries 24, 28, 63, and 70 only.

- 6 Locate the **Element Size** section. Click the **Custom** button.
- 7 Locate the Element Size Parameters section.
- 8 Select the Maximum element size check box. In the associated text field, type 0.002.
- 9 Click 🖷 Build Selected.

# STUDY I

Since it is an elastic case and the external load is proportional, only two load cases are necessary to capture the stress amplitude evaluated in the S-N curve.

Step 1: Stationary

- I In the Model Builder window, expand the Study I node, then click Step I: Stationary.
- 2 In the Settings window for Stationary, click to expand the Study Extensions section.
- **3** Select the **Auxiliary sweep** check box.
- 4 Click + Add.
- **5** In the table, enter the following settings:

Parameter name	Parameter value list
para (Load cycle control)	0 1

6 In the Home toolbar, click **=** Compute.

#### **GLOBAL DEFINITIONS**

Interpolation 1 (int1)

- I In the Home toolbar, click f(X) Functions and choose Global>Interpolation.
- 2 In the Settings window for Interpolation, locate the Definition section.
- 3 From the Data source list, choose File.
- 4 Find the Functions subsection. In the table, enter the following settings:

Function name	Position in file
wohler	1

# 5 Click 📂 Browse.

- 6 Browse to the model's Application Libraries folder and double-click the file bracket\_fatigue\_sn\_curve.txt.
- 7 Click **[]** Import.

#### ADD PHYSICS

- I In the Home toolbar, click 🖄 Add Physics to open the Add Physics window.
- 2 Go to the Add Physics window.
- **3** Find the **Physics interfaces in study** subsection. In the table, clear the **Solve** check box for **Study I**.
- 4 In the tree, select Structural Mechanics>Fatigue (ftg).
- 5 Click Add to Component I in the window toolbar.
- 6 In the Home toolbar, click 🖄 Add Physics to close the Add Physics window.

## FATIGUE (FTG)

Stress-Life 1

- I Right-click **Component I (comp1)>Fatigue (ftg)** and choose the boundary evaluation **Stress-Life**.
- 2 In the Settings window for Stress-Life, locate the Boundary Selection section.
- 3 From the Selection list, choose All boundaries.
- 4 Locate the Fatigue Model Selection section. From the σ list, choose Signed von Mises (principal).
- 5 From the Modification list, choose Stress factor.
- 6 Locate the Solution Field section. From the Physics interface list, choose Solid Mechanics (solid).
- 7 Locate the Fatigue Model Parameters section. From the  $f_{\rm SN}(N)$  list, choose wohler.
- 8 In the k text field, type k.
- 9 Locate the **Evaluation Settings** section. In the  $N_{\rm cut}$  text field, type 1e6.

#### ADD STUDY

- I In the Home toolbar, click  $\sim\sim$  Add Study to open the Add Study window.
- 2 Go to the Add Study window.
- 3 Find the Physics interfaces in study subsection. In the table, clear the Solve check box for Solid Mechanics (solid).
- 4 Find the Studies subsection. In the Select Study tree, select Preset Studies for Selected Physics Interfaces>Fatigue.
- 5 Click Add Study in the window toolbar.
- 6 In the Home toolbar, click  $\sim\sim$  Add Study to close the Add Study window.

#### STUDY 2

#### Step 1: Fatigue

- I In the Settings window for Fatigue, locate the Values of Dependent Variables section.
- 2 Find the Values of variables not solved for subsection. From the Settings list, choose User controlled.
- **3** From the **Method** list, choose **Solution**.
- 4 From the Study list, choose Study I, Stationary.

## Parametric Sweep

- I In the Study toolbar, click **Parametric Sweep**.
- 2 In the Settings window for Parametric Sweep, locate the Study Settings section.
- 3 Click + Add.
- **4** In the table, enter the following settings:

Parameter name	Parameter value list
k (Stress factor)	0.7 0.28

**5** In the **Study** toolbar, click **= Compute**.

# RESULTS

## Surface 1

- I In the Model Builder window, expand the Cycles to Failure (ftg) node, then click Surface I.
- 2 In the Settings window for Surface, click to expand the Quality section.

3 In the Cycles to Failure (ftg) toolbar, click **O** Plot.



To get a better view of the critical points, you can rotate and zoom the bracket. For this purpose, it is useful to have a dedicated View node:

- **4** Click the **5** Show More Options button in the Model Builder toolbar.
- 5 In the Show More Options dialog box, in the tree, select the check box for the node Results>Views.
- 6 Click OK.

View 3D 2

- I In the Model Builder window, under Results right-click Views and choose View 3D.
- 2 Use the mouse to rotate, zoom, and pan until you see the critical point up close.
- 3 In the Settings window for View 3D, locate the View section.
- 4 Select the Lock camera check box.

Now apply this view to the plots.

#### Stress (solid)

- I In the Model Builder window, under Results click Stress (solid).
- 2 In the Settings window for 3D Plot Group, locate the Plot Settings section.
- 3 From the View list, choose View 3D 2.

**4** In the **Stress (solid)** toolbar, click **I** Plot.

Compare the resulting plot with that in Figure 3.

Cycles to Failure (ftg)

- I In the Model Builder window, click Cycles to Failure (ftg).
- 2 In the Settings window for 3D Plot Group, locate the Plot Settings section.
- 3 From the View list, choose View 3D 2.
- 4 In the Cycles to Failure (ftg) toolbar, click Plot.Compare with Figure 4.

Marker I

- I In the Model Builder window, expand the Results>Cycles to Failure (ftg)>Surface I node.
- 2 Right-click Marker I and choose Disable.
- **3** Right-click **Marker I** and choose **Enable**.