

# Tapered Cantilever with Two Load Cases

# Introduction

This example, taken from a NAFEMS benchmark collection (Ref. 1), demonstrates how to apply and evaluate different boundary conditions acting on a cantilever beam.

# Model Definition

The cantilever beam has a thickness of 0.1 m, width of 4 m and is 4 m on its long edge and 2 m on its short edge. Two cases are considered. In the first one a gravity load, mg, acts in the negative y direction with an acceleration of  $9.81 \text{ m/s}^2$ . The left end boundary is fully fixed (no displacements). In the second case gravity load is not present and instead a uniformly distributed horizontal load, F, of 10 MN/m acts along the right end. At the left end there is no displacement in the x direction. Also at a midpoint location the left end is fixed in the y direction; see Figure 1.

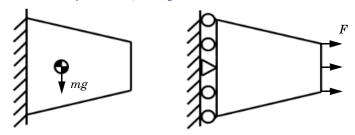


Figure 1: Schematic description of two load cases.

## MATERIAL MODEL

The model uses the following material properties:

- The material is isotropic.
- The Young's modulus (elasticity modulus) is 210 GPa.
- The Poisson ratio is 0.3.
- The density is 7000 kg/m<sup>3</sup>.

# Results and Discussion

For the gravity case shear stress is evaluated. The benchmark target value of -0.200 MPa at the point (0, 2) is in good agreement with the model results. Using the default Normal mesh size, the COMSOL Multiphysics solution gives a value of -0.199 MPa; see Figure 2.

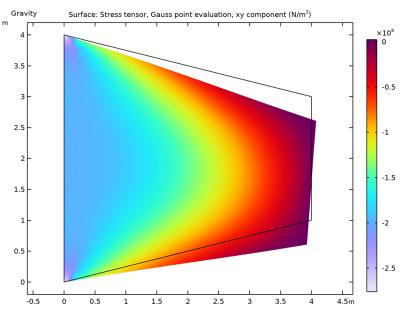


Figure 2: In plane shear stress due to gravity load.

For the load case, the horizontal (x direction) normal stress is evaluated. The benchmark target value of 61.3 MPa at the point (0, 2) is in good agreement with the results. Using the default mesh size, the COMSOL Multiphysics solution gives a value of 61.4 MPa; see Figure 3.

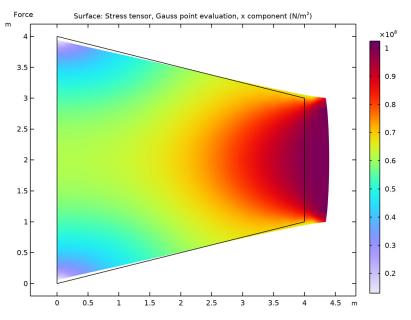


Figure 3: Counteracting stress due to applied edge load.

# Notes About the COMSOL Implementation

Use predefined variables for the acceleration of gravity and density to enter the gravity load as a force/volume. COMSOL then computes the load using the thickness of the geometry.

Use the Solid Mechanics interface to perform a stress analysis. The finite element model uses the default second-order triangular Lagrange elements. To show convergence toward the benchmark value, create a finer mesh.

Use the Load Group and Constraint Group features to collect conditions that are enabled in different studies. Define a group in Global Definitions and assign its content directly from the load or constraint itself. You define load cases in the Study Extensions of a Study Type and enable active load and constraint groups; see Figure 4, where lg denotes a load group and cg denotes a constraint group. Weight is used as a multiplication factor for the corresponding load group.

<b>V</b> Define load case	es					
Load case	IgGravity	Weight	IgForce	Weight	cgGravity	cgForce
Gravity	<ul> <li>✓</li> </ul>	1.0	×	1.0	<ul> <li>Image: A set of the set of the</li></ul>	×
Force	×	1.0	~	1.0	×	~

## Figure 4: Definition of load cases.

## Reference

1. D. Hitchings, A. Kamoulakos, and G.A.O. Davies, *Linear Statics Benchmarks Vol. 1*, NAFEMS, Glasgow, 1987.

**Application Library path:** COMSOL\_Multiphysics/Structural\_Mechanics/ tapered\_cantilever

# Modeling Instructions

From the File menu, choose New.

#### NEW

In the New window, click **Model Wizard**.

#### MODEL WIZARD

- I In the Model Wizard window, click 🤏 2D.
- 2 In the Select Physics tree, select Structural Mechanics>Solid Mechanics (solid).
- 3 Click Add.
- 4 Click 🔿 Study.
- 5 In the Select Study tree, select General Studies>Stationary.
- 6 Click 🗹 Done.

## GEOMETRY I

Polygon I (poll)

I In the Geometry toolbar, click / Polygon.

- 2 In the Settings window for Polygon, locate the Coordinates section.
- 3 From the Data source list, choose Vectors.
- **4** In the **x** text field, type 0 4 4 0 0.
- **5** In the **y** text field, type 0 1 3 4 0.
- 6 Click 틤 Build Selected.

Point I (ptl)

- I In the **Geometry** toolbar, click **Point**.
- 2 In the Settings window for Point, locate the Point section.
- **3** In the **y** text field, type **2**.
- 4 Click 틤 Build Selected.

Form Union (fin)

- I In the Model Builder window, click Form Union (fin).
- 2 In the Settings window for Form Union/Assembly, click 틤 Build Selected.

## MATERIALS

Material I (mat1)

- I In the Model Builder window, under Component I (compl) right-click Materials and choose Blank Material.
- 2 In the Settings window for Material, locate the Material Contents section.
- **3** In the table, enter the following settings:

Property	Variable	Value	Unit	Property group
Young's modulus	E	210[GPa]	Pa	Young's modulus and Poisson's ratio
Poisson's ratio	nu	0.3	I	Young's modulus and Poisson's ratio
Density	rho	7000[kg/m^3]	kg/m³	Basic

## GLOBAL DEFINITIONS

Load Group Gravity

- I In the Model Builder window, right-click Global Definitions and choose Load and Constraint Groups>Load Group.
- 2 In the Settings window for Load Group, type Load Group Gravity in the Label text field.

3 In the **Parameter name** text field, type lgGravity.

#### Load Group Force

- I In the Model Builder window, right-click Load and Constraint Groups and choose Load Group.
- 2 In the Settings window for Load Group, type Load Group Force in the Label text field.
- 3 In the **Parameter name** text field, type lgForce.

#### Constraint Group Gravity

- I In the Model Builder window, right-click Global Definitions and choose Load and Constraint Groups>Constraint Group.
- 2 In the Settings window for Constraint Group, type Constraint Group Gravity in the Label text field.
- 3 In the **Parameter name** text field, type cgGravity.

#### Constraint Group Force

- I Right-click Global Definitions and choose Load and Constraint Groups>Constraint Group.
- 2 In the Settings window for Constraint Group, type Constraint Group Force in the Label text field.
- 3 In the **Parameter name** text field, type cgForce.

#### SOLID MECHANICS (SOLID)

- I In the Model Builder window, under Component I (comp I) click Solid Mechanics (solid).
- 2 In the Settings window for Solid Mechanics, locate the 2D Approximation section.
- 3 From the list, choose Plane stress.
- **4** Locate the **Thickness** section. In the d text field, type **0.1**.

First, define all constraints for both load cases and then assign them to different load and constraint groups.

#### Fixed Constraint I

- I In the Physics toolbar, click Boundaries and choose Fixed Constraint.
- **2** Select Boundaries 1 and 3 only.
- 3 In the Physics toolbar, click () Constraint Group and choose Constraint Group I.

#### Body Load I

- I In the Physics toolbar, click 🔵 Domains and choose Body Load.
- **2** Select Domain 1 only.

- 3 In the Settings window for Body Load, locate the Force section.
- **4** Specify the  $\mathbf{F}_{\mathbf{V}}$  vector as

0	x
-g_const*solid.rho	у

5 In the Physics toolbar, click **Load Group** and choose Load Group Gravity.

Roller I

- I In the Physics toolbar, click Boundaries and choose Roller.
- 2 Select Boundaries 1 and 3 only.
- 3 In the Physics toolbar, click () Constraint Group and choose Constraint Group Force.

#### Boundary Load I

- I In the Physics toolbar, click Boundaries and choose Boundary Load.
- 2 Select Boundary 5 only.
- 3 In the Settings window for Boundary Load, locate the Force section.
- 4 From the Load type list, choose Force per unit length.
- **5** Specify the  $\mathbf{F}_{\mathrm{L}}$  vector as

10[MN/m]	x
0	у

6 In the Physics toolbar, click Load Group and choose Load Group Force.

#### Fixed Constraint 2

- I In the Physics toolbar, click 💭 Points and choose Fixed Constraint.
- 2 Select Point 2 only.

## MESH I

In the Model Builder window, under Component I (comp1) right-click Mesh I and choose Build All.

#### STUDY I

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Step 1: Stationary
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- I In the Model Builder window, under Study I click Step I: Stationary.
- 2 In the Settings window for Stationary, click to expand the Study Extensions section.
- **3** Select the **Define load cases** check box.

4 Click + Add.

**5** In the table, enter the following settings:

Load case	lgGravity	Weight	lgForce	Weight	cgGravity	cgForce
Gravity	$\checkmark$	1.0		1.0	$\checkmark$	

6 Click + Add.

7 In the table, enter the following settings:

Load case	lgGravity	Weight	lgForce	Weight	cgGravity	cgForce
Force		1.0		1.0		$\checkmark$

The study extension for the load cases should look like Figure 4.

8 In the **Home** toolbar, click **= Compute**.

#### RESULTS

#### Normal stress

In the Settings window for 2D Plot Group, type Normal stress in the Label text field.

#### Surface 1

- I In the Model Builder window, expand the Normal stress node, then click Surface I.
- 2 In the Settings window for Surface, click Replace Expression in the upper-right corner of the Expression section. From the menu, choose Component I (compl)>Solid Mechanics> Stress (Gauss points)>Stress tensor, Gauss point evaluation (spatial frame) N/m<sup>2</sup>> solid.sGpx Stress tensor, Gauss point evaluation, x component.
- **3** In the Normal stress toolbar, click **I** Plot.

#### Shear stress

- I In the Model Builder window, right-click Normal stress and choose Duplicate.
- 2 In the Settings window for 2D Plot Group, type Shear stress in the Label text field.
- 3 Locate the Data section. From the Load case list, choose Gravity.

#### Surface 1

- I In the Model Builder window, expand the Shear stress node, then click Surface I.
- 2 In the Settings window for Surface, locate the Expression section.
- **3** In the **Expression** text field, type **solid**.sGpxy.
- **4** In the **Shear stress** toolbar, click **I** Plot.

Point Evaluation - normal stress

- I In the **Results** toolbar, click  $\frac{8.85}{e-12}$  **Point Evaluation**.
- 2 In the Settings window for Point Evaluation, type Point Evaluation normal stress in the Label text field.
- 3 Select Point 2 only.
- 4 Click Replace Expression in the upper-right corner of the Expressions section. From the menu, choose Component I (compl)>Solid Mechanics>Stress (Gauss points)>
   Stress tensor, Gauss point evaluation (spatial frame) N/m<sup>2</sup>>solid.sGpx Stress tensor, Gauss point evaluation, x component.
- 5 Click **= Evaluate**.

Point Evaluation - shear stress

- I In the Results toolbar, click  $\frac{8.85}{e-12}$  Point Evaluation.
- 2 In the **Settings** window for **Point Evaluation**, type Point Evaluation shear stress in the **Label** text field.
- **3** Select Point 2 only.
- 4 Click Replace Expression in the upper-right corner of the Expressions section. From the menu, choose Component I (compl)>Solid Mechanics>Stress (Gauss points)>
   Stress tensor, Gauss point evaluation (spatial frame) N/m<sup>2</sup>>solid.sGpxy Stress tensor, Gauss point evaluation, xy component.
- 5 Click **= Evaluate**.