

# Unsteady 3D Flow Past a Cylinder

## Introduction

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Fluid flow past a cylinder is a common test case in computational fluid dynamics. The flow pattern is characterized by the Reynolds number which is defined as

$$\text{Re} = \frac{\rho U_{\text{mean}} D}{\mu}$$

where  $\rho$  is the density,  $U_{\text{mean}}$  is the mean velocity of the free stream,  $D$  is the cylinder diameter, and  $\mu$  is the dynamic viscosity.

The flow patterns around a cylinder in a free stream for different Reynolds numbers are shown in [Ref. 1](#). At  $\text{Re}$  below 5, the flow remains attached to the cylinder. For  $\text{Re}$  between 5 and 15, steady wake vortices start forming on the downstream side of the cylinder. The wake becomes unsteady and forms a laminar vortex street for  $\text{Re}$  between 40 and 150.

Flow around a cylinder in a channel is even more complicated due to the effect of wall boundaries. Computer simulations of this problem at the intermediate  $\text{Re}$  regime (between 40 and 150) are challenging since they need to be 3D and time-dependent.

In this verification model, a benchmark problem of unsteady, incompressible 3D flow past a cylinder for  $\text{Re} = 100$  during a period of 8 seconds is considered. The lift and drag coefficients are computed and are compared with those in [Ref. 2](#).

## Model Definition

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The geometry is a cylinder of radius  $R$  with the axis parallel to the  $z$ -axis, and placed at  $(x_c, y_c, 0)$ , inside the box  $[0, L] \times [0, H] \times [0, H]$ . [Figure 1](#) shows the geometry corresponding to  $R = 0.05$  m,  $L = 2.5$  m,  $H = 0.5$  m, and  $x_c = 0.5$  m,  $y_c = 0.2$  m.

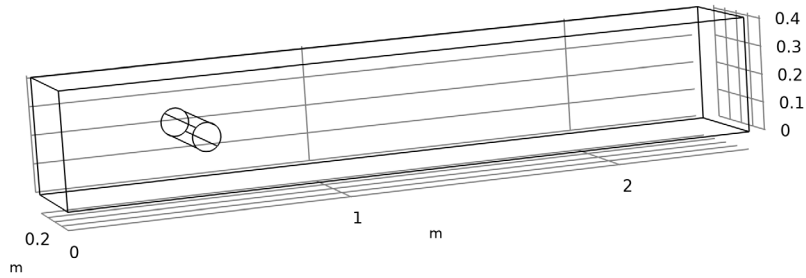
The fluid to be considered is incompressible and Newtonian with a kinematic viscosity of  $10^{-3}$  m<sup>2</sup>/s. The inflow velocity profile varies in time according to

$$U(0, y, z, t) = 36 U_{\text{mean}} yz \frac{(H-y)(H-z)}{H^4} \sin\left(\frac{\pi t}{8}\right), \quad V=W=0.$$

The lift and drag coefficients  $C_D$  and  $C_L$  are computed as functions of time,

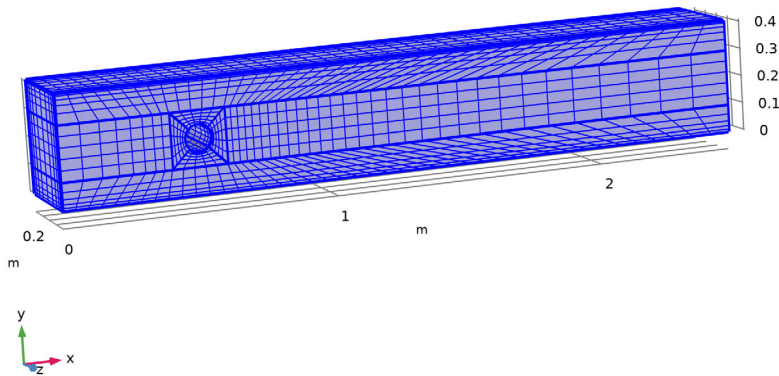
$$C_D(t) = \frac{2F_D(t)}{\rho U_{\text{mean}}^2 A}, \quad C_L(t) = \frac{2F_L(t)}{\rho U_{\text{mean}}^2 A}$$

where  $F_D$  and  $F_L$  are the drag and lift forces, and  $A$  is the projected area,  $A = 2RH$ .



*Figure 1: The geometry is a cylinder placed inside a box.*

The simulation is performed with a relatively coarse mesh with 7200 hexahedral elements shown in [Figure 2](#). P2+P2 shape functions are chosen for the velocity and pressure to allow for better conservation and higher accuracy compared to P2+P1 and P1+P1. The generalized alpha method with automatic time stepping is chosen since it has less damping than the BDF method.



*Figure 2: The relatively coarse mesh, with 7200 hexahedral elements, used in the simulation.*

## Results and Discussion

Figure 3 shows the flow pattern at  $t = 7.95$  s, the last saved time before the inflow velocity returns to zero.

The lift and drag coefficients versus time are shown in Figure 4 and Figure 5 respectively. They capture the general shape of those published Ref. 2 quite well.

When the mesh size is reduced by a factor of 2, resulting in 57,600 elements, the computational time increases by a factor of 8 but the agreement with the current simulation is still excellent.

Time=7.95 s    Slice: Velocity magnitude (m/s)    Streamline: Velocity field    Surface: 1 (1)

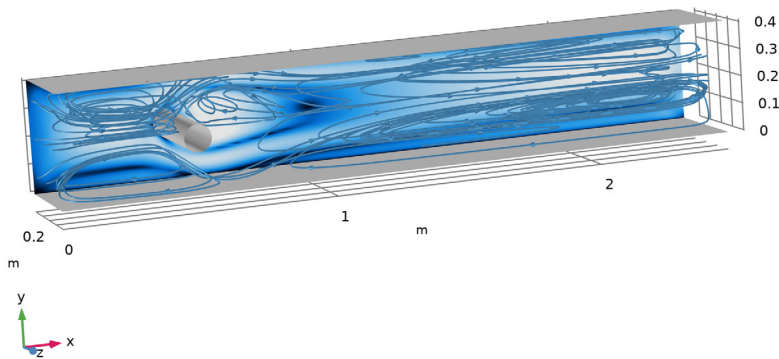


Figure 3: Computed velocity field at  $t = 7.95$  s.

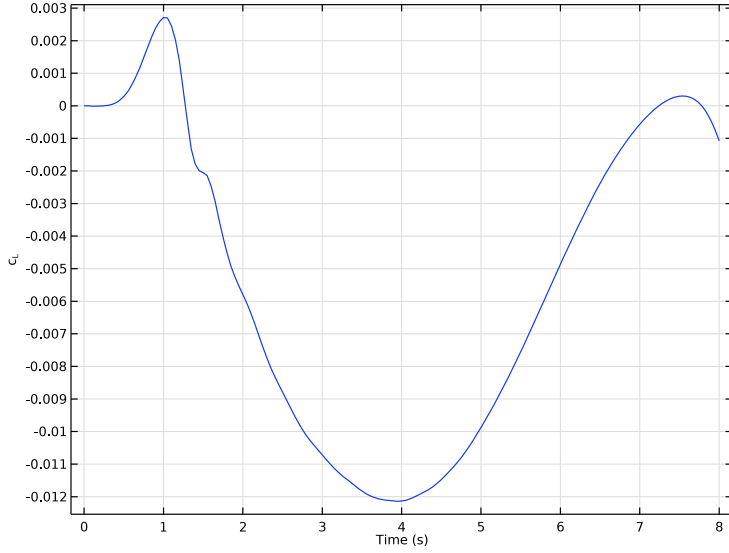


Figure 4: Lift coefficient versus time.

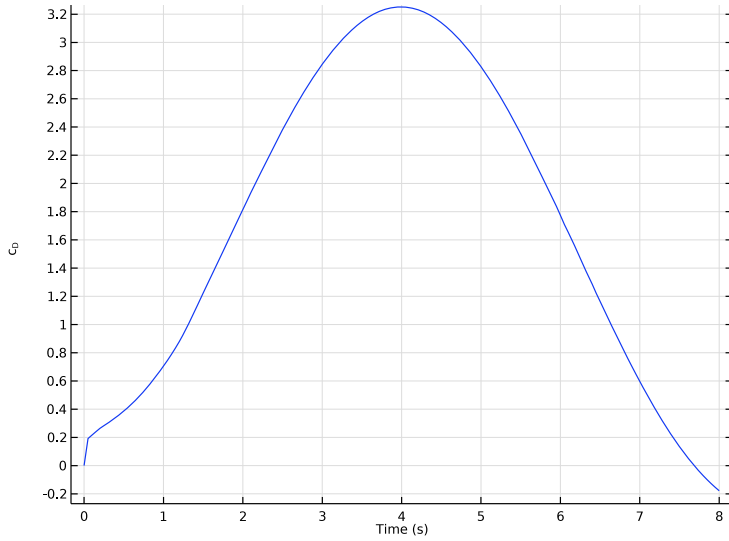


Figure 5: Drag coefficient versus time.

## Notes About the COMSOL Implementation

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The space discretization P2+P2 coupled with the generalized alpha time discretization works efficiently for this application. P2+P2 elements allow for a coarser mesh, better conservation, and more accuracy compared to P2+P1 and P1+P1 elements. The generalized alpha method has less damping than the BDF method. Automatic time-stepping works well and relatively large time steps can be used, and thus less computational time is needed compared to [Ref. 2](#).

## References

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1. M. Van Dyke, *An album of fluid motion*, the Parabolic Press, ISBN 0-915760-03-7, 1982.
2. E. Bayraktar, O. Mierka, and S. Turek, “Benchmark Computation of 3D Laminar Flow Around a Cylinder with CFX, OpenFOAM and FeatFlow,” *IJCSE*, vol. 7, no. 3, pp. 253–266, 2012.

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**Application Library path:** CFD\_Module/Verification\_Examples/  
cylinder\_flow\_3d\_periodic


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## Modeling Instructions




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From the **File** menu, choose **New**.

### NEW

In the **New** window, click  **Model Wizard**.

### MODEL WIZARD

- 1 In the **Model Wizard** window, click  **3D**.
- 2 In the **Select Physics** tree, select **Fluid Flow>Single-Phase Flow>Laminar Flow (spf)**.
- 3 Click **Add**.
- 4 Click  **Study**.
- 5 In the **Select Study** tree, select **General Studies>Time Dependent**.
- 6 Click  **Done**.

## GLOBAL DEFINITIONS

### Parameters 1


- 1 In the **Model Builder** window, under **Global Definitions** click **Parameters 1**.
- 2 In the **Settings** window for **Parameters**, locate the **Parameters** section.
- 3 In the table, enter the following settings:

Name	Expression	Value	Description
U_mean	1[m/s]	1 m/s	Mean inflow velocity
rho	1[kg/m^3]	1 kg/m <sup>3</sup>	Density
mu	0.001[Pa*s]	0.001 Pa·s	Dynamic viscosity
H	0.41[m]	0.41 m	Height and Width
L	2.5[m]	2.5 m	Length
xc	0.5[m]	0.5 m	Cylinder x-pos
yc	0.2[m]	0.2 m	Cylinder y-pos
R	0.05[m]	0.05 m	Cylinder radius

## GEOMETRY 1


First, create the box  $[0, L] \times [0, H] \times [0, H]$ .

### Block 1 (blk1)

- 1 In the **Geometry** toolbar, click  **Block**.
- 2 In the **Settings** window for **Block**, locate the **Size and Shape** section.
- 3 In the **Width** text field, type L.
- 4 In the **Depth** text field, type H.
- 5 In the **Height** text field, type H.

Next, create a smaller box around the cylinder. This box will be used later on in the meshing sequence.



### Block 2 (blk2)

- 1 In the **Geometry** toolbar, click  **Block**.
- 2 In the **Settings** window for **Block**, locate the **Size and Shape** section.
- 3 In the **Width** text field, type  $4 \cdot R$ .
- 4 In the **Depth** text field, type  $4 \cdot R$ .
- 5 In the **Height** text field, type H.
- 6 Locate the **Position** section. In the **x** text field, type  $x_c - 2 \cdot R$ .

7 In the **y** text field, type  $yc - 2 * R$ .



Now, create the cylinder.

#### *Cylinder 1 (cyl1)*



- 1 In the **Geometry** toolbar, click  **Cylinder**.
- 2 In the **Settings** window for **Cylinder**, locate the **Size and Shape** section.
- 3 In the **Radius** text field, type R.
- 4 In the **Height** text field, type H.
- 5 Locate the **Position** section. In the **x** text field, type  $xc$ .
- 6 In the **y** text field, type  $yc$ .
- 7 Locate the **Rotation Angle** section. In the **Rotation** text field, type 45.
- 8 In the **Geometry** toolbar, click  **Build All**.

The following operations divide the flow domain into a number of subdomains. This way, a coarser mesh can be used far from the cylinder.



#### *Line Segment 1 (ls1)*

- 1 In the **Geometry** toolbar, click  **More Primitives** and choose **Line Segment**.
- 2 On the object **cyl1**, select Point 2 only.
- 3 In the **Settings** window for **Line Segment**, locate the **Endpoint** section.
- 4 Find the **End vertex** subsection. Click to select the  **Activate Selection** toggle button.
- 5 On the object **blk2**, select Point 5 only.

#### *Line Segment 2 (ls2)*





- 1 In the **Geometry** toolbar, click  **More Primitives** and choose **Line Segment**.
- 2 On the object **cyl1**, select Point 6 only.
- 3 In the **Settings** window for **Line Segment**, locate the **Endpoint** section.
- 4 Find the **End vertex** subsection. Click to select the  **Activate Selection** toggle button.
- 5 On the object **blk2**, select Point 7 only.

#### *Line Segment 3 (ls3)*


- 1 In the **Geometry** toolbar, click  **More Primitives** and choose **Line Segment**.
- 2 On the object **cyl1**, select Point 8 only.
- 3 In the **Settings** window for **Line Segment**, locate the **Endpoint** section.
- 4 Find the **End vertex** subsection. Click to select the  **Activate Selection** toggle button.
- 5 On the object **blk2**, select Point 8 only.




#### Line Segment 4 (ls4)

- 1 In the **Geometry** toolbar, click  **More Primitives** and choose **Line Segment**.
- 2 On the object **cyll**, select Point 4 only.
- 3 In the **Settings** window for **Line Segment**, locate the **Endpoint** section.
- 4 Find the **End vertex** subsection. Click to select the  **Activate Selection** toggle button.
- 5 On the object **blk2**, select Point 6 only.
- 6 Click  **Build All Objects**.
- 7 Click the  **Wireframe Rendering** button in the **Graphics** toolbar, and rotate the geometry to get a better view.

#### Block 3 (blk3)




- 1 In the **Geometry** toolbar, click  **Block**.
- 2 In the **Settings** window for **Block**, locate the **Size and Shape** section.
- 3 In the **Width** text field, type L.
- 4 In the **Depth** text field, type 2\*R.
- 5 In the **Height** text field, type H.


#### Block 4 (blk4)

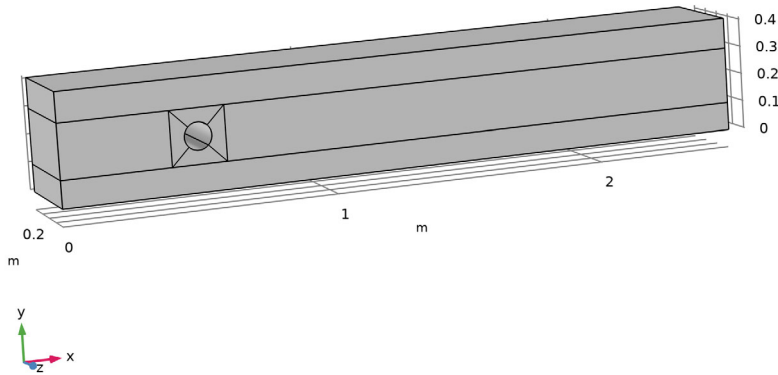
- 1 Right-click **Block 3 (blk3)** and choose **Duplicate**.
- 2 In the **Settings** window for **Block**, locate the **Size and Shape** section.
- 3 In the **Depth** text field, type H-6\*R.
- 4 Locate the **Position** section. In the **y** text field, type 6\*R.
- 5 Click  **Build All Objects**.

Now, create the final computational domain with a hollow cylinder.


#### Difference 1 (dif1)

- 1 In the **Geometry** toolbar, click  **Booleans and Partitions** and choose **Difference**.
- 2 Select the objects **blk1** and **blk2** only.
- 3 In the **Settings** window for **Difference**, locate the **Difference** section.
- 4 Find the **Objects to subtract** subsection. Click to select the  **Activate Selection** toggle button.
- 5 Select the object **cyll** only.
- 6 In the **Geometry** toolbar, click  **Build All**.

- 7 Click the  **Wireframe Rendering** button in the **Graphics** toolbar. The geometry looks like the following image.



### LAMINAR FLOW (SPF)


- 1 Click the  **Show More Options** button in the **Model Builder** toolbar.
- 2 In the **Show More Options** dialog box, in the tree, select the check box for the node **Physics>Stabilization**.
- 3 Click **OK**.
- 4 In the **Model Builder** window, under **Component 1 (comp1)** click **Laminar Flow (spf)**.
- 5 In the **Settings** window for **Laminar Flow**, click to expand the **Consistent Stabilization** section.
- 6 Find the **Navier-Stokes equations** subsection. Clear the **Crosswind diffusion** check box.
- 7 Click to expand the **Discretization** section. From the **Discretization of fluids** list, choose **P2+P2**. P2+P2 is used because it is more conservative and more accurate than P2+P1 and P1+P1.

#### *Fluid Properties I*

- 1 In the **Model Builder** window, under **Component 1 (comp1)>Laminar Flow (spf)** click **Fluid Properties I**.

- 2 In the **Settings** window for **Fluid Properties**, locate the **Fluid Properties** section.
- 3 From the  $\rho$  list, choose **User defined**. In the associated text field, type  $\rho$ .
- 4 From the  $\mu$  list, choose **User defined**. In the associated text field, type  $\mu$ .

#### *Inlet 1*

- 1 In the **Physics** toolbar, click  **Boundaries** and choose **Inlet**.
- 2 Select Boundaries 1, 5, and 9 only.
- 3 In the **Settings** window for **Inlet**, locate the **Velocity** section.
- 4 In the  $U_0$  text field, type  $36*U\_mean*z*y*(H-y)*(H-z)/H^4*\sin(\pi*t/8[s])$ .

Here,  $1/[s]$  is used to make the input of  $\sin()$  dimensionless.

#### *Outlet 1*

- 1 In the **Physics** toolbar, click  **Boundaries** and choose **Outlet**.
- 2 Select Boundaries 31–33 only.

### **MESH 1**

Use advanced operations such as Map and Sweep to create a hexahedral mesh.

#### *Mapped 1*

In the **Mesh** toolbar, click  **Boundary** and choose **Mapped**.


#### *Size*

- 1 In the **Model Builder** window, click **Size**.
- 2 In the **Settings** window for **Size**, locate the **Element Size** section.
- 3 From the **Calibrate for** list, choose **Fluid dynamics**.
- 4 From the **Predefined** list, choose **Coarse**.

#### *Mapped 1*

- 1 In the **Model Builder** window, click **Mapped 1**.
- 2 Select Boundaries 17, 18, 20, and 25 only.

#### *Distribution 1*


- 1 Right-click **Mapped 1** and choose **Distribution**.
- 2 Select Edges 22, 24, 28, 32, 33, 36, 39, and 46 only.
- 3 In the **Settings** window for **Distribution**, click  **Build Selected**.

#### *Distribution 2*


- 1 Right-click **Mapped 1** and choose **Distribution**.

- 2 Select Edges 23 and 27 only.
- 3 In the **Settings** window for **Distribution**, locate the **Distribution** section.
- 4 From the **Distribution type** list, choose **Predefined**.
- 5 In the **Element ratio** text field, type 2.
- 6 From the **Growth rate** list, choose **Exponential**.

#### *Distribution 3*

- 1 Right-click **Mapped 1** and choose **Distribution**.
- 2 Select Edges 40 and 42 only.
- 3 In the **Settings** window for **Distribution**, locate the **Distribution** section.
- 4 From the **Distribution type** list, choose **Predefined**.
- 5 In the **Element ratio** text field, type 2.
- 6 From the **Growth rate** list, choose **Exponential**.
- 7 Click  **Build All**.

#### *Mapped 2*

- 1 In the **Mesh** toolbar, click  **Boundary** and choose **Mapped**.
- 2 Select Boundaries 8 and 29 only.

#### *Distribution 1*

- 1 Right-click **Mapped 2** and choose **Distribution**.
- 2 Select Edges 9 and 56 only.

#### *Distribution 2*


- 1 In the **Model Builder** window, right-click **Mapped 2** and choose **Distribution**.
- 2 Select Edges 10 and 15 only.
- 3 In the **Settings** window for **Distribution**, locate the **Distribution** section.
- 4 From the **Distribution type** list, choose **Predefined**.
- 5 In the **Number of elements** text field, type 8.
- 6 In the **Element ratio** text field, type 3.
- 7 From the **Growth rate** list, choose **Exponential**.

#### *Distribution 3*

- 1 Right-click **Mapped 2** and choose **Distribution**.
- 2 Select Edges 47 and 50 only.
- 3 In the **Settings** window for **Distribution**, locate the **Distribution** section.

- 4 From the **Distribution type** list, choose **Predefined**.
- 5 In the **Number of elements** text field, type 30.
- 6 In the **Element ratio** text field, type 6.
- 7 From the **Growth rate** list, choose **Exponential**.
- 8 Select the **Reverse direction** check box.

#### *Mapped 3*

- 1 In the **Mesh** toolbar, click  **Boundary** and choose **Mapped**.
- 2 Select Boundaries 4 and 12 only.

#### *Distribution 1*

- 1 Right-click **Mapped 3** and choose **Distribution**.
- 2 Select Edges 14 and 59 only.
- 3 In the **Settings** window for **Distribution**, locate the **Distribution** section.
- 4 From the **Distribution type** list, choose **Predefined**.
- 5 In the **Element ratio** text field, type 4.
- 6 From the **Growth rate** list, choose **Exponential**.


#### *Distribution 2*

- 1 In the **Model Builder** window, right-click **Mapped 3** and choose **Distribution**.
- 2 Select Edges 4 and 53 only.
- 3 In the **Settings** window for **Distribution**, locate the **Distribution** section.
- 4 From the **Distribution type** list, choose **Predefined**.
- 5 In the **Element ratio** text field, type 4.
- 6 From the **Growth rate** list, choose **Exponential**.
- 7 Select the **Reverse direction** check box.


#### *Distribution 3*

- 1 Right-click **Mapped 3** and choose **Distribution**.
- 2 Select Edges 5 and 18 only.
- 3 In the **Settings** window for **Distribution**, locate the **Distribution** section.
- 4 From the **Distribution type** list, choose **Predefined**.
- 5 In the **Number of elements** text field, type 43.
- 6 In the **Element ratio** text field, type 1.6.
- 7 From the **Growth rate** list, choose **Exponential**.

### *Swept 1*

In the **Mesh** toolbar, click  **Swept**.

### *Distribution 1*

- 1 Right-click **Swept 1** and choose **Distribution**.
- 2 In the **Settings** window for **Distribution**, locate the **Distribution** section.
- 3 From the **Distribution type** list, choose **Predefined**.
- 4 In the **Number of elements** text field, type 10.
- 5 In the **Element ratio** text field, type 4.
- 6 From the **Growth rate** list, choose **Exponential**.
- 7 Select the **Symmetric distribution** check box.
- 8 Click  **Build All**.

The mesh in [Figure 2](#) is now generated.


## **STUDY 1**

### *Step 1: Time Dependent*


- 1 In the **Model Builder** window, under **Study 1** click **Step 1: Time Dependent**.
- 2 In the **Settings** window for **Time Dependent**, locate the **Study Settings** section.
- 3 In the **Output times** text field, type range (0, 0.05, 8).
- 4 From the **Tolerance** list, choose **User controlled**.
- 5 In the **Relative tolerance** text field, type 0.001.

### *Solution 1 (sol1)*

Choose the generalized alpha method for the time stepping. It has less damping than the BDF.

- 1 In the **Study** toolbar, click  **Show Default Solver**.
- 2 In the **Model Builder** window, expand the **Solution 1 (sol1)** node.
- 3 In the **Model Builder** window, under **Study 1>Solver Configurations>Solution 1 (sol1)** click **Time-Dependent Solver 1**.
- 4 In the **Settings** window for **Time-Dependent Solver**, click to expand the **Time Stepping** section.
- 5 From the **Method** list, choose **Generalized alpha**.
- 6 Select the **Initial step** check box.
- 7 In the associated text field, type 0.01.

8 From the **Maximum step constraint** list, choose **Constant**.

9 In the **Study** toolbar, click  **Compute**.

Evaluate the drag and lift coefficients.

## RESULTS

### Surface 2

1 In the **Results** toolbar, click  **More Datasets** and choose **Surface**.

2 Select Boundaries 21–24 only.


### Integral 1

1 In the **Results** toolbar, click  **More Datasets** and choose **Evaluation>Integral**.

2 In the **Settings** window for **Integral**, locate the **Data** section.

3 From the **Dataset** list, choose **Surface 2**.

### Point Evaluation 1

1 In the **Results** toolbar, click  **Point Evaluation**.

2 In the **Settings** window for **Point Evaluation**, locate the **Data** section.

3 From the **Dataset** list, choose **Integral 1**.

4 Locate the **Expressions** section. In the table, enter the following settings:

Expression	Unit	Description
$-\text{spf.T\_stressy}/(0.5*\text{spf.rho}*(U\_mean)^2*(2*R*H))$	1	Lift coefficient
$-\text{spf.T\_stressx}/(0.5*\text{spf.rho}*(U\_mean)^2*(2*R*H))$	1	Drag coefficient

5 Click  **Evaluate**.

## TABLE

1 Go to the **Table** window.

2 Click **Table Graph** in the window toolbar.

## TABLE

1 Go to the **Table** window.

2 Click **Table Graph** in the window toolbar.

Plot the lift coefficient versus time as shown in [Figure 4](#).

3 In the **Model Builder** window, under **Results>ID Plot Group 3** click **Table Graph 1**.

- 4 In the **Settings** window for **Table Graph**, locate the **Data** section.
- 5 From the **Plot columns** list, choose **Manual**.
- 6 In the **Columns** list, select **Lift coefficient (I), Integral I**.

## RESULTS

### *Lift coefficient*


- 1 In the **Model Builder** window, under **Results** click **ID Plot Group 3**.
- 2 In the **Settings** window for **ID Plot Group**, type **Lift coefficient** in the **Label** text field.
- 3 Locate the **Plot Settings** section. Select the **y-axis label** check box.
- 4 In the associated text field, type  $c_L$ .

### *Drag coefficient*

Plot the drag coefficient versus time as shown in [Figure 5](#).

- 1 Right-click **Lift coefficient** and choose **Duplicate**.
- 2 In the **Settings** window for **ID Plot Group**, type **Drag coefficient** in the **Label** text field.
- 3 Locate the **Plot Settings** section. In the **y-axis label** text field, type  $c_D$ .



### *Table Graph 1*

- 1 In the **Model Builder** window, expand the **Drag coefficient** node, then click **Table Graph 1**.
- 2 In the **Settings** window for **Table Graph**, locate the **Data** section.
- 3 In the **Columns** list, select **Drag coefficient (I), Integral I**.
- 4 In the **Drag coefficient** toolbar, click  **Plot**.

### *Point Evaluation 1*

Evaluate the maximum and minimum of the coefficients.

### *Point Evaluation 2*



- 1 In the **Model Builder** window, under **Results>Derived Values** right-click **Point Evaluation 1** and choose **Duplicate**.
- 2 In the **Settings** window for **Point Evaluation**, locate the **Data Series Operation** section.
- 3 From the **Transformation** list, choose **Maximum**.
- 4 Click  next to  **Evaluate**, then choose **New Table**.



## TABLE

Go to the **Table** window.

### *Point Evaluation 3*

- 1 Right-click **Point Evaluation 2** and choose **Duplicate**.
- 2 In the **Settings** window for **Point Evaluation**, locate the **Expressions** section.
- 3 Click to select row number 2 in the table.
- 4 Locate the **Data Series Operation** section. From the **Transformation** list, choose **Minimum**.
- 5 Click  next to  **Evaluate**, then choose **Table 2 - Point Evaluation 2**.

### *Surface 3*

- 1 In the **Results** toolbar, click  **More Datasets** and choose **Surface**.
- 2 Select Boundaries 2, 13, and 21–24 only.

Now, visualize the velocity field as shown in [Figure 3](#).

### *Velocity (spf)*

- 1 In the **Model Builder** window, under **Results** click **Velocity (spf)**.
- 2 In the **Settings** window for **3D Plot Group**, locate the **Color Legend** section.
- 3 Clear the **Show legends** check box.
- 4 Locate the **Plot Settings** section. Clear the **Plot dataset edges** check box.
- 5 Locate the **Data** section. From the **Time (s)** list, choose **7.95**. Since the inlet velocity vanishes at the final time step, the solution at  $t=7.95\text{s}$  is chosen for a better visualization of the streamlines.

### *Slice*

Create a slice in the middle of the computational domain, parallel to the  $xy$  – plane to see the flow pattern more clearly.

- 1 In the **Model Builder** window, expand the **Velocity (spf)** node, then click **Slice**.
- 2 In the **Settings** window for **Slice**, locate the **Plane Data** section.
- 3 From the **Plane** list, choose **xy-planes**.
- 4 In the **Planes** text field, type 1.
- 5 Locate the **Coloring and Style** section. From the **Color table** list, choose **JupiterAuroraBorealis**.

### *Deformation 1*

Shift the slice back to the wall to get a better view.

- 1 Right-click **Slice** and choose **Deformation**.
- 2 In the **Settings** window for **Deformation**, locate the **Expression** section.
- 3 In the **z component** text field, type  $-0.205$ .
- 4 Locate the **Scale** section. Select the **Scale factor** check box.
- 5 In the associated text field, type 1.

#### *Streamline 1*

Create and add arrows to the streamlines.

- 1 In the **Model Builder** window, right-click **Velocity (spf)** and choose **Streamline**.
- 2 In the **Settings** window for **Streamline**, locate the **Streamline Positioning** section.
- 3 In the **Number** text field, type 10.
- 4 Select Boundaries 1, 5, and 9 only.
- 5 Locate the **Coloring and Style** section. Find the **Line style** subsection. From the **Type** list, choose **Tube**.
- 6 Select the **Radius scale factor** check box.
- 7 In the associated text field, type  $0.003$ .
- 8 Find the **Point style** subsection. From the **Type** list, choose **Arrow**.
- 9 From the **Color** list, choose **Custom**.
- 10 On Windows, click the colored bar underneath, or — if you are running the cross-platform desktop — the **Color** button.
- 11 Click **Define custom colors**.
- 12 Set the RGB values to 71, 145, and 199, respectively.
- 13 Click **Add to custom colors**.
- 14 Click **Show color palette only** or **OK** on the cross-platform desktop.

#### *Surface 1*

- 1 Right-click **Velocity (spf)** and choose **Surface**.
- 2 In the **Settings** window for **Surface**, locate the **Data** section.
- 3 From the **Dataset** list, choose **Surface 3**.
- 4 Locate the **Expression** section. In the **Expression** text field, type 1.
- 5 Locate the **Coloring and Style** section. From the **Coloring** list, choose **Uniform**.
- 6 From the **Color** list, choose **Gray**.