

Maximizing the Buckling Load of a Diagonal Brace

Introduction

Buckling is a common failure mode for slender structures. One can distinguish between tensile and compressive modes, but it is rare to see buckling failure due to a tensile load. The buckling load can normally be increased by adding material, but oftentimes it is possible to achieve the same effect by relocating material.

Model Definition

The model studies a structural member, which provides diagonal stiffness in a larger structure. The buckling load of the member is maximized by changing two geometrical parameters, the thickness and the width. The case of a member with a length of 4 m as well as 6 m are studied. The member has a slender geometry as illustrated in Figure 1.



x y z

Figure 1: The computational domain is symmetric, but this is not exploited in the model.

The member is made of steel, and the thickness is small, so its structural response can be well characterized using a shell model. The brace is put in compression using a **Rigid Connector** boundary condition with an **Applied Force** feature. The initial critical load factor and volume is first calculated and then a second study dealing with the optimization is set up.

Results

The result of the optimization for a length of 6 m is shown in Figure 2. The Buckling load has been increased by 140%. The bound of the volume constraint is scaled with the brace length, so the 6 m brace is 50% heavier than the 4 m brace. This causes the optimization result for a 4 m brace to have a smaller width and larger thickness than the 6 m brace.

Lz=4, Lxy=0.082151, thickness=0.0030543 Critical load factor=20.919 Surface: Displacement magnitude (m)

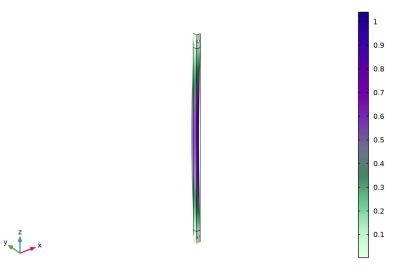


Figure 2: The optimized 6 m member is shown for the 1st buckling mode.

Application Library path: Optimization_Module/Design_Optimization/ diagonal_brace_buckling_optimization

Modeling Instructions

From the File menu, choose New.

N E W In the New window, click 💇 Model Wizard.

MODEL WIZARD

- I In the Model Wizard window, click 间 3D.
- 2 In the Select Physics tree, select Structural Mechanics>Shell (shell).
- 3 Click Add.
- 4 Click \bigcirc Study.
- 5 In the Select Study tree, select Preset Studies for Selected Physics Interfaces> Linear Buckling.
- 6 Click **M** Done.

GLOBAL DEFINITIONS

Parameters 1

- I In the Model Builder window, under Global Definitions click Parameters I.
- 2 In the Settings window for Parameters, locate the Parameters section.
- **3** In the table, enter the following settings:

Name	Expression	Value	Description
Lz	6[m]	6 m	Length of brace
Lxy	5[cm]	0.05 m	Brace width
thickness	5[mm]	0.005 m	Brace thickness
Lz1	3*Lxy	0.15 m	Brace end length
Rhole	Lz1/10	0.015 m	Hole radius
Fload	1[kN]	1000 N	Load

GEOMETRY I

Block I (blkI)

- I In the **Geometry** toolbar, click 🗍 **Block**.
- 2 In the Settings window for Block, locate the Size and Shape section.
- **3** In the **Width** text field, type Lxy.
- 4 In the **Depth** text field, type Lxy.
- **5** In the **Height** text field, type Lz.

6 Click to expand the Layers section. In the table, enter the following settings:

Layer name	Thickness (m)
Layer 1	Lz1
Layer 1	Lz-2*Lz1

Boundaries to Delete

- I In the Geometry toolbar, click 🖓 Selections and choose Box Selection.
- 2 In the Settings window for Box Selection, type Boundaries to Delete in the Label text field.
- 3 Locate the Geometric Entity Level section. From the Level list, choose Boundary.
- 4 Locate the **Box Limits** section. In the **x minimum** text field, type Lxy*0.1.
- **5** In the **y minimum** text field, type Lxy*0.1.

Delete Entities 1 (del1)

- I In the Model Builder window, right-click Geometry I and choose Delete Entities.
- 2 In the Settings window for Delete Entities, locate the Entities or Objects to Delete section.
- **3** From the Selection list, choose Boundaries to Delete.

Cylinder I (cyl1)

- I In the **Geometry** toolbar, click 问 **Cylinder**.
- 2 In the Settings window for Cylinder, locate the Size and Shape section.
- **3** In the **Radius** text field, type Rhole.
- 4 In the **Height** text field, type Rhole.
- **5** Locate the **Position** section. In the **x** text field, type Lxy/2.
- 6 In the y text field, type -Rhole/2.
- 7 In the z text field, type Lz1/2.
- 8 Locate the Axis section. From the Axis type list, choose y-axis.

Cylinder 2 (cyl2)

- I Right-click Cylinder I (cyll) and choose Duplicate.
- 2 In the Settings window for Cylinder, locate the Position section.
- **3** In the **z** text field, type Lz Lz1/2.

Difference I (dif1)

- I In the Geometry toolbar, click i Booleans and Partitions and choose Difference.
- 2 Select the object dell only.

- 3 In the Settings window for Difference, locate the Difference section.
- **4** Find the **Objects to subtract** subsection. Select the **Delivate Selection** toggle button.
- 5 Select the objects cyll and cyl2 only.
- 6 Click the 🔛 Show Grid button in the Graphics toolbar.

The geometry should now look like that in Figure 1.

Internal Edges

- I In the Geometry toolbar, click 🗞 Selections and choose Box Selection.
- 2 In the Settings window for Box Selection, type Internal Edges in the Label text field.
- 3 Locate the Geometric Entity Level section. From the Level list, choose Edge.
- 4 Locate the Box Limits section. In the z minimum text field, type Lz1*0.999.
- **5** In the **z maximum** text field, type Lz1*1.001.
- 6 Locate the Output Entities section. From the Include entity if list, choose Entity inside box.

Cylinder Selection 1 (cylsel1)

- I In the Geometry toolbar, click 🖓 Selections and choose Cylinder Selection.
- 2 In the Settings window for Cylinder Selection, locate the Geometric Entity Level section.
- **3** From the Level list, choose Edge.
- 4 Locate the Size and Shape section. In the Outer radius text field, type Rhole*1.01.
- **5** Locate the **Position** section. In the **x** text field, type Lxy/2.
- **6** In the z text field, type Lz1/2.
- 7 Locate the Axis section. From the Axis type list, choose y-axis.

Cylinder Selection 2 (cylsel2)

- I Right-click Cylinder Selection I (cylsell) and choose Duplicate.
- 2 In the Settings window for Cylinder Selection, locate the Position section.
- **3** In the **z** text field, type Lz Lz1/2.

ADD MATERIAL

- I In the Home toolbar, click 🙀 Add Material to open the Add Material window.
- 2 Go to the Add Material window.
- 3 In the tree, select Built-in>Structural steel.
- 4 Click Add to Component in the window toolbar.
- 5 In the Home toolbar, click 🙀 Add Material to close the Add Material window.

SHELL (SHELL)

Thickness and Offset I

- I In the Model Builder window, under Component I (compl)>Shell (shell) click Thickness and Offset I.
- 2 In the Settings window for Thickness and Offset, locate the Thickness and Offset section.
- **3** In the *d* text field, type thickness.

Rigid Connector 1

- I In the Physics toolbar, click 🔚 Edges and choose Rigid Connector.
- 2 In the Settings window for Rigid Connector, locate the Edge Selection section.
- **3** From the Selection list, choose Cylinder Selection I.
- **4** Locate the **Prescribed Displacement at Center of Rotation** section. Select the **Prescribed in x direction** check box.
- **5** Select the **Prescribed in y direction** check box.
- 6 Select the Prescribed in z direction check box.
- 7 Locate the Prescribed Rotation section. From the By list, choose Constrained rotation.
- 8 Select the Constrain rotation around z-axis check box.
- 9 Select the Constrain rotation around x-axis check box.

Rigid Connector 2

- I Right-click Rigid Connector I and choose Duplicate.
- 2 In the Settings window for Rigid Connector, locate the Edge Selection section.
- **3** From the Selection list, choose Cylinder Selection **2**.
- **4** Locate the **Prescribed Displacement at Center of Rotation** section. Clear the **Prescribed in z direction** check box.

Applied Force 1

- I In the Physics toolbar, click 🧮 Attributes and choose Applied Force.
- 2 In the Settings window for Applied Force, locate the Applied Force section.
- **3** Specify the **F** vector as

0	x
0	у
-Fload	z

MESH I

Edge I

- I In the Mesh toolbar, click \bigwedge Boundary and choose Edge.
- 2 In the Settings window for Edge, locate the Edge Selection section.
- 3 From the Selection list, choose Internal Edges.

Distribution I

Right-click Edge I and choose Distribution.

Mapped I

- I In the Mesh toolbar, click \bigwedge Boundary and choose Mapped.
- **2** Select Boundaries 3 and 4 only.

Distribution I

- I Right-click Mapped I and choose Distribution.
- 2 Select Edge 4 only.
- 3 In the Settings window for Distribution, locate the Distribution section.
- 4 In the Number of elements text field, type 20.

Free Triangular 1

- I In the Mesh toolbar, click \triangle Boundary and choose Free Triangular.
- 2 In the Settings window for Free Triangular, locate the Boundary Selection section.
- 3 From the Geometric entity level list, choose Remaining.

Size

- I In the Model Builder window, click Size.
- 2 In the Settings window for Size, locate the Element Size section.
- **3** From the **Predefined** list, choose **Extra fine**.
- 4 Click 📗 Build All.

DEFINITIONS

- I Click the 🐱 Show More Options button in the Model Builder toolbar.
- 2 In the Show More Options dialog box, select General>Variable Utilities in the tree.
- 3 In the tree, select the check box for the node General>Variable Utilities.
- 4 Click OK.

Mass Properties 1 (mass1)

- I In the Model Builder window, under Component I (compl) right-click Definitions and choose Variable Utilities>Mass Properties.
- 2 In the Settings window for Mass Properties, locate the Source Selection section.
- **3** From the **Geometric entity level** list, choose **Boundary**.
- **4** From the **Selection** list, choose **All boundaries**.
- 5 Locate the Density section. From the Density source list, choose From physics interface.

INITIAL DESIGN

- I In the Model Builder window, click Study I.
- 2 In the Settings window for Study, type Initial Design in the Label text field.
- **3** In the **Home** toolbar, click **= Compute**.

RESULTS

Mass and Critical Load Factor

- I In the **Results** toolbar, click **Let Evaluation Group**.
- 2 In the **Settings** window for **Evaluation Group**, type Mass and Critical Load Factor in the **Label** text field.

Global Evaluation 1

- I Right-click Mass and Critical Load Factor and choose Global Evaluation.
- 2 In the Settings window for Global Evaluation, click Add Expression in the upper-right corner of the Expressions section. From the menu, choose Component I (comp1)> Definitions>Mass Properties 1>mass1.mass Mass kg.
- 3 Click Add Expression in the upper-right corner of the Expressions section. From the menu, choose Component I (comp1)>Shell>shell.LFcrit Critical load factor.
- **4** In the Mass and Critical Load Factor toolbar, click **=** Evaluate.

Mode Shape (shell), Shell Geometry (shell), Thickness and Orientation (shell)

- I In the Model Builder window, under Results, Ctrl-click to select Mode Shape (shell), Shell Geometry (shell), and Thickness and Orientation (shell).
- 2 Right-click and choose Group.

Initial Design

In the **Settings** window for **Group**, type Initial Design in the **Label** text field.

GLOBAL DEFINITIONS

Parameters 1

- I In the Model Builder window, under Global Definitions click Parameters I.
- 2 In the Settings window for Parameters, locate the Parameters section.

3 In the table, enter the following settings:

Name	Expression	Value	Description
МО	23.5[kg]	23.5 kg	Initial Mass

ADD STUDY

- I In the Home toolbar, click $\sim\sim$ Add Study to open the Add Study window.
- 2 Go to the Add Study window.
- 3 Find the Studies subsection. In the Select Study tree, select Preset Studies for Selected Physics Interfaces>Linear Buckling.
- 4 Click Add Study in the window toolbar.
- 5 In the Home toolbar, click ~ 2 Add Study to close the Add Study window.

STUDY 2

Parametric Sweep

- I In the Study toolbar, click **Parametric Sweep**.
- 2 In the Settings window for Parametric Sweep, locate the Study Settings section.
- 3 Click + Add.
- 4 In the table, enter the following settings:

Parameter name	Parameter value list	Parameter unit
Lz (Length of brace)	4 6	m

- 5 In the Model Builder window, click Study 2.
- 6 In the Settings window for Study, type Optimization in the Label text field.

Optimization

- I In the Study toolbar, click optimization.
- 2 In the Settings window for Optimization, locate the Optimization Solver section.
- 3 From the Method list, choose COBYLA.

4 Locate the **Objective Function** section. In the table, enter the following settings:

Expression	Description	Evaluate for
abs(comp1.shell.LFcrit)		Linear Buckling

This means that both compressive and tensile failure modes are considered.

- 5 From the Type list, choose Maximization.
- 6 From the Solution list, choose Minimum of objectives.
- 7 Locate the Control Variables and Parameters section. Click + Add twice.
- 8 In the table, enter the following settings:

Parameter name	Initial value	Scale	Lower bound	Upper bound
Lxy (Brace width)	5[cm]	5[cm]	1[cm]	20[cm]
thickness (Brace thickness)	5[mm]	5[mm]	1 [mm]	20[mm]

9 Click Add Expression in the upper-right corner of the Constraints section. From the menu, choose Component I (compl)>Definitions>Mass Properties l>compl.massl.mass - Mass - kg.

IO Locate the **Constraints** section. In the table, enter the following settings:

Expression	Lower bound	Upper bound	Evaluate for
comp1.mass1.mass/ MO		Lz/(6[m])	Linear Buckling

II In the **Study** toolbar, click **= Compute**.

RESULTS

Mode Shape (shell) I, Shell Geometry (shell) I, Thickness and Orientation (shell) I

- In the Model Builder window, under Results, Ctrl-click to select Mode Shape (shell) I, Shell Geometry (shell) I, and Thickness and Orientation (shell) I.
- 2 Right-click and choose Group.

Optimization

- I In the Settings window for Group, type Optimization in the Label text field.
- 2 Right-click Optimization and choose Plot.
- **3** Click the **- Zoom Extents** button in the **Graphics** toolbar.

Global Evaluation 2

- I In the Model Builder window, under Results>Mass and Critical Load Factor right-click Global Evaluation I and choose Duplicate.
- 2 In the Settings window for Global Evaluation, locate the Data section.
- 3 From the Dataset list, choose Optimization/Parametric Solutions I (sol5).
- **4** In the Mass and Critical Load Factor toolbar, click **=** Evaluate.

The critical load factor has been improved with a factor of 2 for the 6 m brace.

Create a new plot for the model thumbnail.

Thumbnail

- I In the Home toolbar, click 🚛 Add Plot Group and choose 3D Plot Group.
- 2 In the Settings window for 3D Plot Group, type Thumbnail in the Label text field.
- 3 Locate the Data section. From the Dataset list, choose Shell I.

Surface 1

Right-click Thumbnail and choose Surface.

Deformation I

In the Model Builder window, right-click Surface I and choose Deformation.

Surface 2

- I In the Model Builder window, under Results>Thumbnail right-click Surface I and choose Duplicate.
- 2 In the Settings window for Surface, locate the Data section.
- 3 From the Dataset list, choose Shell 2.
- 4 Click to expand the Inherit Style section. From the Plot list, choose Surface I.
- **5** Click the **for the XY View** button in the **Graphics** toolbar.

Deformation I

- I In the Model Builder window, expand the Surface 2 node, then click Deformation I.
- 2 In the Settings window for Deformation, locate the Expression section.
- 3 In the x component text field, type u-4.
- 4 Right-click Deformation I and choose Copy.

Line I

- I In the Model Builder window, right-click Thumbnail and choose Line.
- 2 In the Settings window for Line, locate the Data section.

- 3 From the Dataset list, choose Shell 2.
- **4** Locate the **Expression** section. In the **Expression** text field, type **1**.
- 5 Locate the Coloring and Style section. From the Coloring list, choose Uniform.
- 6 From the Color list, choose Black.
- 7 Click to expand the Inherit Style section. From the Plot list, choose Surface I.
- 8 Clear the **Color** check box.

Deformation I

- I In the Model Builder window, right-click Line I and choose Paste Deformation.
- 2 In the Thumbnail toolbar, click **I** Plot.
- **3** Click the **Zoom Extents** button in the **Graphics** toolbar.

14 | MAXIMIZING THE BUCKLING LOAD OF A DIAGONAL BRACE